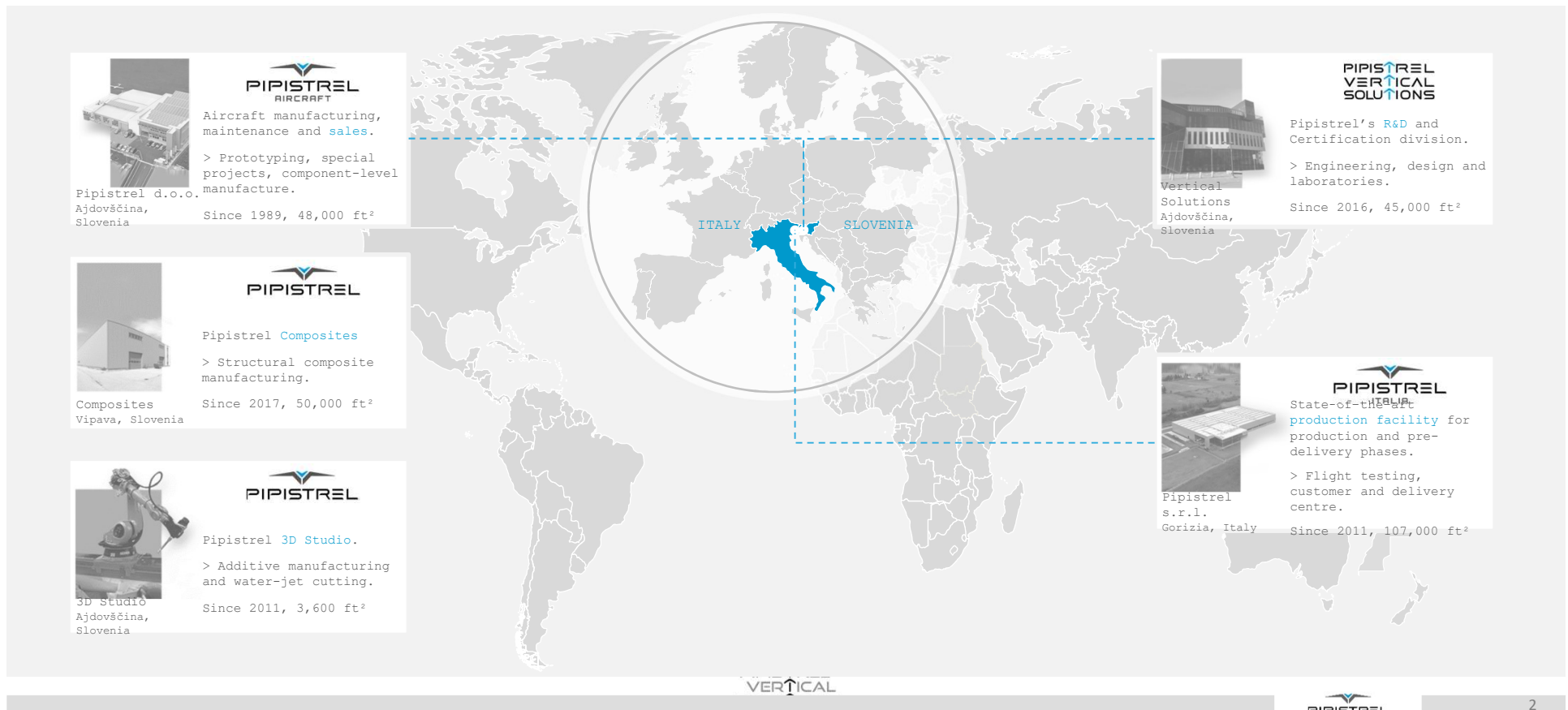


The Future of Flight Technology

Tine Tomazic, CTO | Nov 2022

The company has created a lean, scalable, end-to-end design and production capability, while ensuring ongoing innovation with world-leading partners



It is moving to a vertically-integrated model to ensure a seamless, end-to-end experience for customers



Pipistrel's capabilities cover the full idea-to-certification process in-house

Production capacity will increase to 685,000 ft², or 450 aircraft per year

The sales network now includes 53 dealers and contract manufacturers

Pipistrel has partnered with 64 flight schools worldwide

- 1 CONCEPTUAL DESIGN**
3D CAD models, project phases, estimated budget.
- 2 AERODYNAMIC ANALYSIS & DESIGN**
Including computational fluid dynamics.
- 3 STRUCTURAL ANALYSIS & DESIGN**
Predicting materials' behaviour by load condition.
- 4 SYSTEMS DESIGN**
Design and integration of aircraft systems.
- 5 ELECTRIC / HYBRID POWERTRAINS**
Design and airframe integration of powertrains.
- 6 AVIONICS AND ELECTRONICS**
Custom electronics boards, software, or interfaces.
- 7 STRUCTURAL TESTING**
Testing of full-scale airframes and aircraft components.
- 8 FLIGHT TESTING**
Test plan, data collection and analysis.
- 9 PROTOTYPE CONSTRUCTION**
Construction and tooling, including composites.
- CERTIFICATION AND STC**
Certification with EASA or supplementary type certs.



1 DEALER NETWORK TARGETS CONSUMERS AND FLIGHT SCHOOL CUSTOMERS

Dealers by region

2 DEDICATED SALES TEAM TARGETS OEM, CORPORATE AND GOVERNMENT CUSTOMERS

PIPISTREL | OEM | Corporate | Government

3 LOCALISED MANUFACTURING & DISTRIBUTION PARTNERS TO PENETRATE NEW REGIONS

PIPISTREL ASIA-PACIFIC

WORLDWIDE NETWORK OF FLIGHT SCHOOLS PROVIDES TRAINING BUT ALSO MARKETING

Flight schools by region

DEALERS PROVIDE 1ST LINE MAINTENANCE, WHILE PIPISTREL HQ PROVIDES SPECIALIST SUPPORT

DEALERS | General maintenance

PIPISTREL ASIA-PACIFIC | Maintenance and parts

PIPISTREL | Specialist support and parts

EASA TC and TC-in-progress

EASA.A.573

SW 121A



Velis Club



Velis Electro



EASA.E.234

E-811/810 Electric Engine



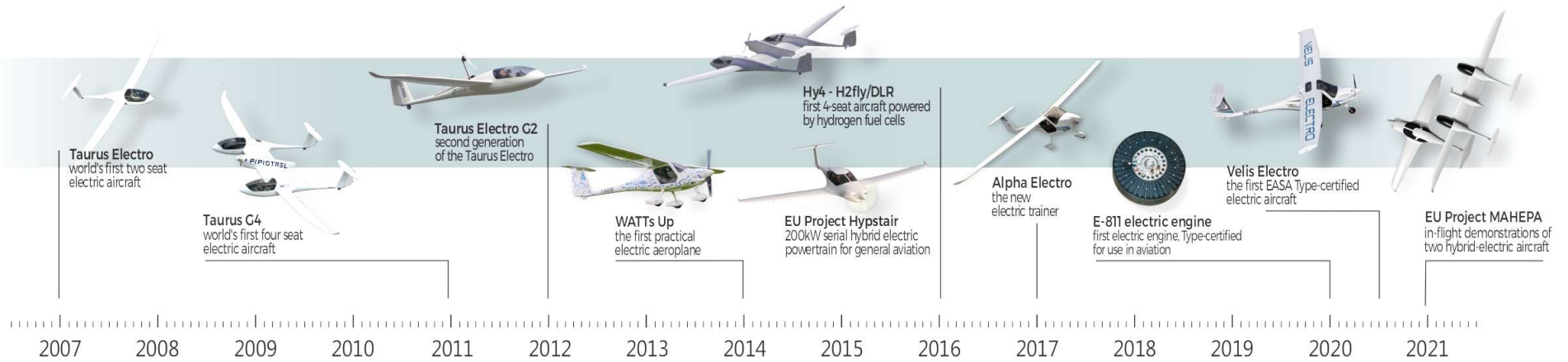
EASA.A.660

Panthera 152





PIPISTREL



Flying electrically since 2007.

Research - MAHEPA

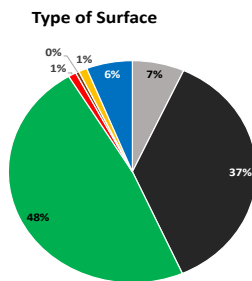
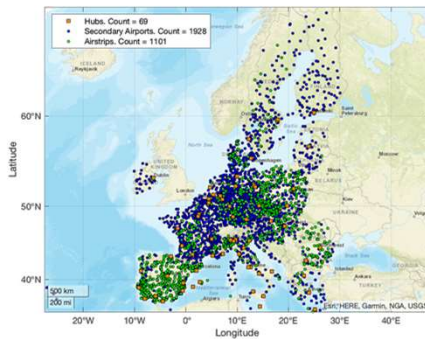


Photography: Jozef Kovacic - MAHEPA



MARKET STUDY

AVAILABLE AERODROME NETWORK



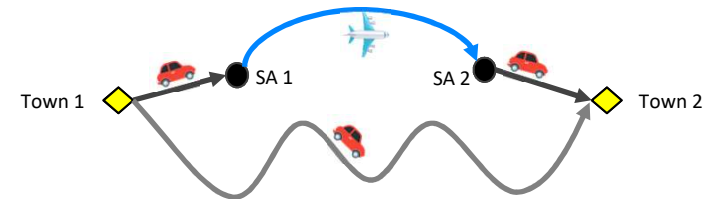
- >> **69** hubs
- 3027** secondary aerodromes
- >> Runway length ≥ 800 m; **50%** total airports
- >> **90%** secondary aerodromes has an airport within **100 km**

TRAVEL SCENARIOS

MICROFEEDER



MINILINER



CRITERIUM: time advantage of selected transport



PRVK-X

Increased complexity,
optimised for DOC

PRVK-1

Optimised for low noise, short
runways, steep arrival/departure
H2B-direct turbine compatible

PRVK-LR

Long range, high speed
H2B-direct turbine compatible



True innovators never consider failure, only the consequences of success

↑ UNIFIER19 kick-off meeting (October 2019):

↑ „...*the **only responsible, sensible and sustainable** way for European aviation for **regional** air-traffic to embrace, is going towards **complete zero emission flight.**“*

↑ European Commission, INEA and Clean Sky workshop (January 2020):

↑ „***The project’s focus** will be, from now on, on **hydrogen and battery based propulsion** for the well-being of European citizens and leadership of European Aerospace Industry.“*




PIPISTREL




PIPISTREL

UNIFIER 19 COMMUNITY FRIENDLY MINILINER



69 + 3000

Connected with **3000 small airports** in EU



- ↑ Bring air-travel closer to people without compromising their well being
- ↑ Zero emission, quiet flight
- ↑ Minimal infrastructure investment



EPCA H2 demo concepts

ENVIRONMENTAL OBJECTIVES

CO₂ AND NO_x EMISSION REDUCTION

- >> 100% reduction
- >> Fuel-cell hybrid system with liquid hydrogen tank
- >> We do not consider water, or water vapors as a harmful emission due to lower operational altitudes

NOISE EMISSION REDUCTION

- >> >10 dB reduction with respect to reference aircraft

REFERENCE AIRCRAFT

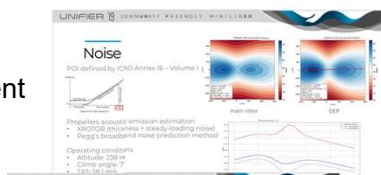
- >> Twin turbo prop >> Optimised for Unifier19 mission
- >> State-of-the-art propulsion, aerodynamics and structural characteristics
- >> Within CS-23 regulation requirements



MULTIDISCIPLINARY OPTIMISATION

>> C7A: Traditional and DEP, WTP and BLI

>> Noise assessment



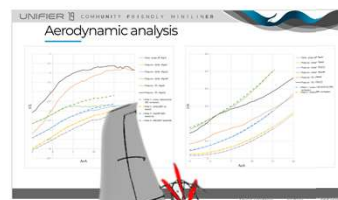
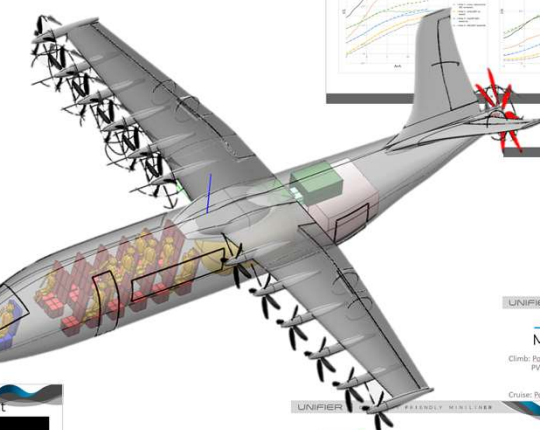
>> Propeller design



>> Qualitative structural assessment

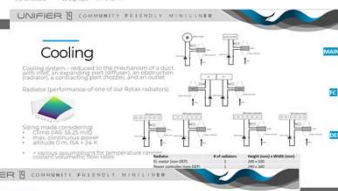
Qualitative structural assessment

Section	Pros	Cons
Fuselage	Manufacture flexibility, robust, proven	Limited opportunity for innovation
Main Wing	New composite geometry, simplified attachment to the fuselage	Insufficient payload capacity if wing compromised the design
Engine	Proven traditional methods could be applied	Added complexity, lengthy design powerplant integration requires
Landing gear	Evenly distributed weight of electric engines acts as load alleviation for structural engineering load case	Lack of off-axis due to the presence of electric engines may affect the landing gear
Overall	Conventionally proven configuration	Relatively simple, easy to manufacture, very conventional and historically proven design

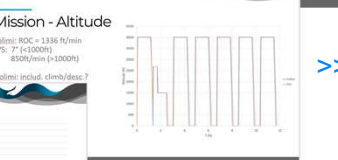


>> Aerodynamic analysis

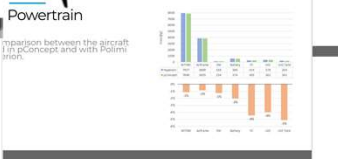
>> Cooling drag prediction



>> Mission analysis



>> Powertrain sizing



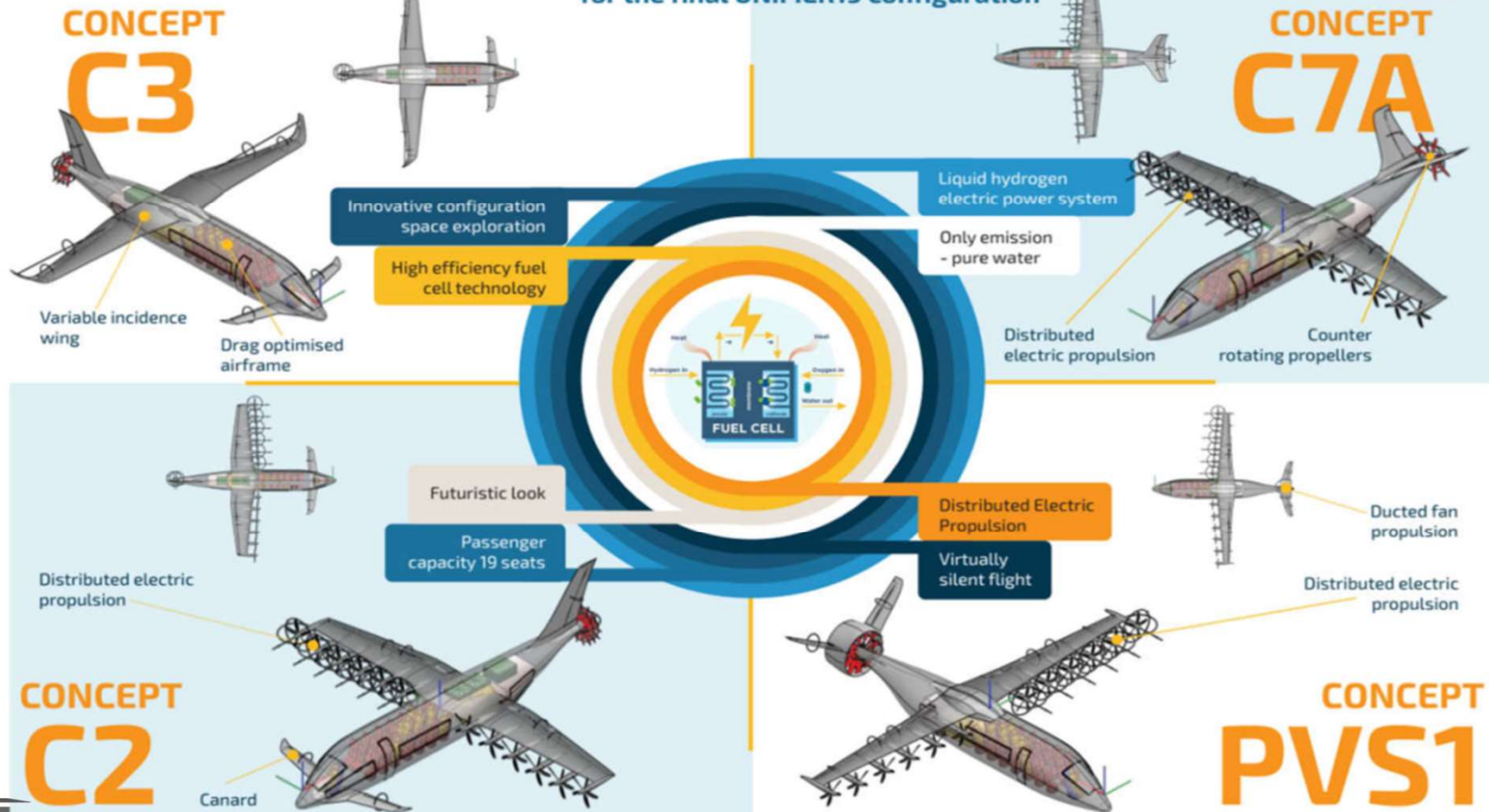
>> Operating costs estimation

Operating Costs

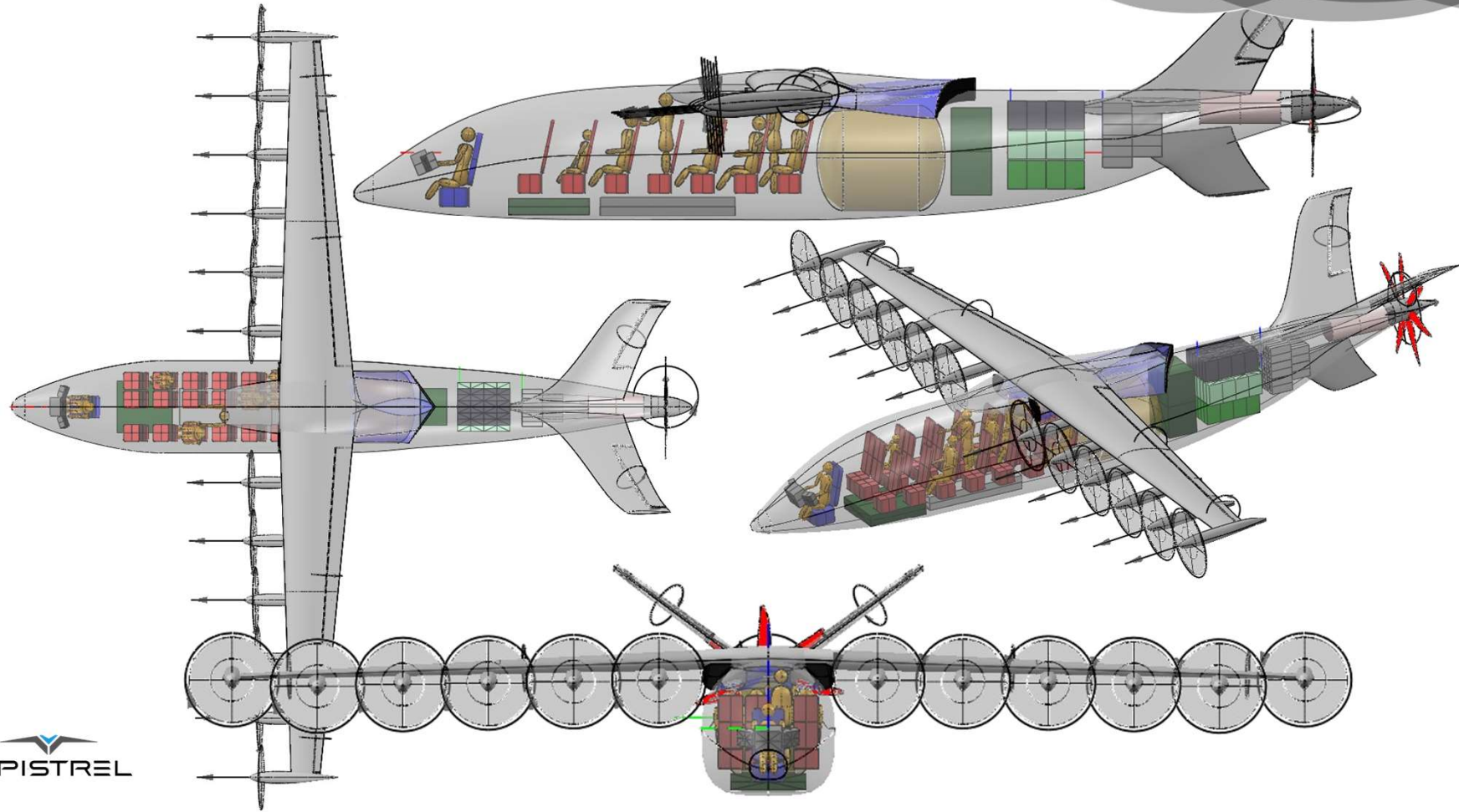
	UNIFIER 19	C7A
Total DOC per flight	141.64	112.72
DOC per hour	24.80	20.29
DOC per km per seat (0.24)	4.80	4.06
DOC per km per seat (0.24)	4.80	4.06
DOC per km	12.00	10.15
DOC per km	4.80	4.06

DOC is relatively high as cruise speed is low (143 KTAS @ 4000 ft) → all time-based costs increase (crew, overhaul, utilisation).

Four competing candidates have been downselected for the final UNIFIER19 configuration



UNIFIER 19 COMMUNITY FRIENDLY MINILINER




PIPISTREL

UNIFIER 19 COMMUNITY FRIENDLY MINILINER




PIPISTREL

Concepts overview

↑ ~2 MW of total installed power – Synergies with regional-sized airplanes

- 1-1.5 MW from Fuel Cells
- 1 MW from Battery
- Single LH₂ tank, 250-320 kg capacity
- Compatibility with H2B – propulsion agnostic airframes

↑ Operations from 800 m runways

↑ 250 KTAS, 24000 ft

↑ 1000 NM + 45 min reserves

↑ Fit within CS-23 bounds of today

PRVK-1

Main Characteristics

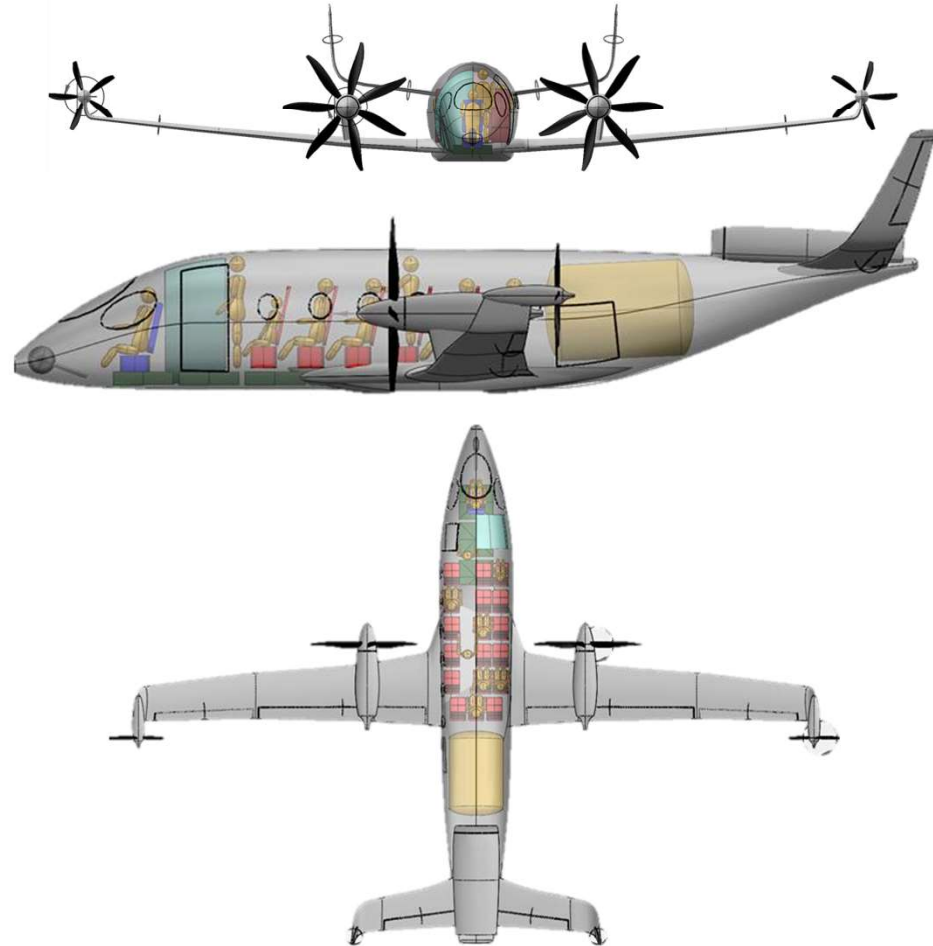
Crew	one pilot
Capacity	19 pax (w/o lavatory)
Length	14.75 m
Wingspan	21.60 m
Height	4.80 m
OEW	5300 kg (w/ battery)
MTOM	7500 kg

Propulsion system

- 1+ MW fuel cell system, works with existing 1500 W/kg, sized for 2200-2500 W/kg on system level
- ~5 m³ Liquid Hydrogen cryogenic tank, ~ 300 kg of LH2
- 1 MW battery, 1500 VDC bus, 480 kg, 2000+ W/kg,
- 2 low speed/low noise tractor propellers (0.35 M tip velocity) driven by twin-motors
- 2 wingtip pusher propellers (single motors)

Typical Mission Performance

Cruise speed	250 KTAS
Range	4x 350 km hops (each hop incl. 100 km diversion + 45 min)
Takeoff distance	< 800 m



Research - FLAGSHIP SPRINGBOARD - Clean Aviation 2022-2028

H2 Aircraft Concepts

Architectures and Target Definition


Concept a/c

Reference a/c



Commuter

Design mission: 19 PAX, 500 km range, cruise speed 500 km/h

- Highly efficient wing
- 2 LH₂ tanks behind PAX cabin - added weight: 0.5 tons
- Distributed propulsion using electric motors for thrust




1. Major assumptions: 20% gravimetric index of LH₂ tank, 90% available LH₂ fuel, FCS mass 1.5 kWh/kg (incl. cooling) and 10% peak eff.
2. Cost per engine and fuel burner
3. Maximum take-off weight


Regional

Design mission: 80 PAX, 1,000 km range, cruise speed Mach 0.44

- Highly efficient wing
- 2 LH₂ tanks behind PAX cabin - added weight: 2 tons
- Distributed propulsion using electric motors for thrust



1. Major assumptions: 20% gravimetric index of LH₂ tank, 90% available LH₂ fuel, FCS mass 1.7 kWh/kg (incl. cooling) and 10% peak eff.
2. Cost per engine and fuel burner
3. Maximum take-off weight




6 "pods"
fuel cell as standalone turboprop motors

Distributed hydrogen
fuel cell propulsion system

8-bladed propellers
made of composite materials

Removable fixtures
allowing wing for quick pool access & disassembly



<100
Passengers


Hydrogen Hybrid Turboprop Engines (x 2)

1,000+nm
Range

Liquid Hydrogen Storage & Distribution System

SMR

Evolutionary aircraft design for short range
Reference aircraft: Airbus A320 neo




Balanced cent of gravity - wt shifted to back increased wing loading

-4% Decrease of block energy due to higher energy efficiency of fuel cell system

Source: DLR design study, expert panel, project team


Evolutionary aircraft design for short range
Reference aircraft: Airbus A320 neo



2 LH₂ tanks integrated in back of fuselage - extension of fuselage by 5m (43m total length)
H2 direct burning turbine used

+6% Increase of block energy due to higher operating empty weight

Source: DLR design study, expert panel, project team




<200
Passengers

Hydrogen Hybrid Turbofan Engines (x 2)

2,000+nm
Range

Liquid Hydrogen Storage & Distribution System

AIRBUS



<100
Passengers

Hydrogen Hybrid Turboprop Engines (x 2)

1,000+nm
Range

Liquid Hydrogen Storage & Distribution System



PIPISTREL



- ↑ Drive the change towards new services with drones in all airspaces, anytime and anywhere
- ↑ Support development of new procedures, technologies and services
- ↑ Deliver environmentally friendly, digital and connected aerial cargo delivery solutions across Europe
- ↑ Being a catalyst of modernization of European air traffic management

Miniliner™ is for:

- Operators who believe in true zero-emission regional and sub-regional aerial mobility
- Cost effective multi-hop and/or long-range operations; 40% TOC reduction in cost from twin turboprops
- Powered by hydrogen fuel cells, using LH₂ 300 kg-class storage and MW-class high-power density long-life battery
- Distributed electric propulsion w/ redundancy for CS-25 (airliner)-level safety and reliability
- 0.35 M tip-speed ultra low-noise propellers
- Fully digital aircraft set-up for single pilot operations from small airfields

